



On-orbit Calibration of the Geostationary Imaging Fourier Transform Spectrometer (GIFTS)

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The Year 2002 Conference on Characterization and Radiometric Calibration for Remote Sensing

> April 29 - May 2, 2002 Space Dynamics Laboratory / Utah State University







The NASA New Millennium Program's Geostationary Imaging Fourier Transform Spectrometer (GIFTS) requires highly accurate radiometric and spectral calibration in order to carry out its mission to provide water vapor, wind, temperature, and trace gas profiling from geostationary orbit. A calibration approach has been developed for the formulation phase GIFTS instrument design. The in-flight calibration is performed using views of two on-board blackbody sources along with cold space. A radiometric calibration uncertainty analysis has been developed and used to show that the expected performance for GIFTS exceeds its top level requirement to measure brightness temperature to better than 1 K.

The spectral calibration is established by the highly stable diode laser used as the reference for interferogram sampling, and verified with comparisons to atmospheric calculations. Algorithms that normalize spectral sampling have been developed to correct for off-axis effects. A baseline approach defining the role that ground measurements play in the on-orbit calibration has been established.







Combine Advanced Measurement Technologies On a Geosynchronous Satellite to Obtain 4-D Observations of the Atmosphere



- Horizontal: Large detector arrays give near instantaneous wide
 2-D geographical coverage
- Vertical: Michelson interferometer (FTS) gives high spectral resolution that yields high vertical resolution
- Temporal: Geosynchronous orbit allows high time resolution (i.e., motion observations)

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Items to be addressed quantitatively in this talk.

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Radiometric Calibration





- Top level requirements
- In-flight Radiometric Calibration Concept for GIFTS
- Predicted Radiometric Performance
- In-flight Calibrator Baseline Design
- Heritage



GIFTS Blackbody Top Level Requirements



Calibration Accuracy [From the GIRD §5.5.1 (GIR398)]

The absolute calibration accuracy of the GIFTS Instrument shall be =1K brightness temperature for Earth scene brightness temperatures >240K (SW/MW channel) and >190K (LW channel) and shall be traceable to the National Institute of Standards and Technology (NIST). The reproducibility shall be =0.2K for the same conditions.

On-Board Calibration [From the GIRD §5.5.3 (GIR400)]

The GIFTS instrument shall provide three calibration reference sources that shall be viewable at regular, programmable intervals. Two of these sources shall be blackbodies at temperatures chosen to optimize calibration accuracy over the life of the mission. The instrument shall include the capability of viewing space as a third calibration source...

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- Two small reference Blackbodies located behind telescope, combined with Space View.
- Blackbody design is scaled from the UW groundbased AERI and NAST / S-HIS aircraft instruments.
- Constraints on original S/C prevented traditional external large aperture blackbody configuration.
- Advantages compared to large external blackbody:
 (1) higher emissivity is practical with small size
 (2) protection from solar forcing.





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GIFTS Radiometric Calibration Concept

$$N = \left(\frac{\boldsymbol{t}_{m}}{\boldsymbol{t}_{t}}\right) (B_{H} - B_{C}) Re\left(\frac{C_{E} - C_{S}}{C_{H} - C_{C}}\right) + B_{S}$$

- Radiance *N* derived from raw spectra of Earth (C_E), Space (C_S), and the internal Hot (C_H) and Cold (C_C) Blackbodies
- *t_t* is the signal transmission of the telescope mirrors &
 t_m is the transmission of the Blackbody pick-off mirror
- B is the Planck Radiance from the Hot, Cold, and Space References

GIFTS Radiometric Calibration Parameters

Absolute Calibration Uncertainties

Input Parameters			Parameter Uncertainties	
wn	1000	Wavenumber, [cm-1]		
TauTel	0.913	Transmission of Telescope (2) plus front flat (1)	² TauRatio	0.0020
TauBBmirror	0.970	Transmission of BB mirror	² TauBBmirror	0.010
Thbb	300	Temp. of Hot Blackbody, [K]	² Thbb	0.1
Tcbb	265	Temp. of Cold Blackbody, [K]	² Tcbb	0.1
Tspace	4.0	Temp. of Space View, [K]		
Ttel	265	Temp. of Front Telescope, [K]	² Ttel	2.00
Tbbmirror	220	Temp. of blackbody mirror, [K]	² Tbbmirror	2.00
Tstr	250	Temp. of Structure Reflecting into BB's, [K]	² Tstr	5.00
Ehbb	0.996	Emissivity of HBB, [-]	²Ehbb	0.002
Ecbb	0.996	Emissivity of CBB, [-]	² Ecbb	0.002
Espace	1.0	Effective Emissivity of Space View, [-]		
Parameters Used For Temp				
TtelChange	0.4	Change in Telescope Temp Between Earth & Space Views		
TbbmirChange	2.0	Change in BB Mirror Temp Between Hot and Cold Views		

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GIFTS Calibration Accuracy-Longwave

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GIFTS Calibration Accuracy-Shortwave

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The derived In-flight calibrator subsystem requirements are:

- Temperature knowledge:
- Emissivity knowledge:
- Temperature gradient :

 ± 0.1 K better than $\pm 0.2\%$ knowledge within 0.1 K

GIFTS Instrument imposed requirements and allocations:

•	Source Aperture:	2.54 cm
•	Envelope:	8.0 cm Dia. X 14 cm long
•	Operating Temperature:	265 to 300 K
•	Mass (2 BB's and Controllers):	< 2.4 kg
•	Power (2 BB's and Controller):	< 3.0 W

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Paint surface is Chemglaze Z306

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GIFTS BB Block Diagram

- UW Developed AERI (groundbased) and S-HIS (aircraft) FTIR Instruments have demonstrated Radiometric Performance with accuracies better than 1 K.
- These programs have demonstrated the methodologies of integrating instrument Radiometric Models with Calibration Techniques and Error Budgets to produce the required radiometric accuracies.

AERI / NIST Reference Blackbody Comparison

Miami IR Workshop 2-4 March 1998

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- @ 333K, Max Error <0.05K@ 303K, Max Error <0.04K
 - @ 293K, Max Error <0.02K

- In-flight radiometric calibration makes use of two small internal blackbody cavities (located near the field stop), plus a space view.
- A combination of the internal blackbodies and the space view allow tracking of any in-flight changes of the fore-optics transmission.
- Internal high-emissivity Blackbodies offer:
 - Better absolute calibration
 - Easier implementation smaller range of pointing mirror angles

Spectral Calibration

Philosophy

- Pre-flight spectral calibration parameters determined during ground calibration.
- Highly stable laser serves as an in-flight calibration reference.
- Verify in-flight using known atmospheric absorption lines.

Discussion Topics

- In-flight wavenumber scale verification.
- The Instrument Line Shape (ILS) correction.
- Off-axis wavenumber scale renormalization.

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- The GIFTS spectral scale is referenced to the on-board laser used to trigger interferogram sampling, as in a normal FTS.
- In a Michelson inteferometer, sampling at equal intervals in optical path delay leads (after Fourier Transform) to equal sampling in wavenumber.
- The actual wavenumber sampling interval depends on both the wavelength of the laser used to sample the interferogram and the angle between the path of the laser and the IR beam for each detector.

GIFTS Spectral Calibration Requirements

- The effective radiometric errors associated with spectral calibration are included under the overall 1 K absolute calibration specification.
- An objective for GIFTS is to provide a laser stability better than 1 part in 10^6 (3 σ) over a month time period.
- » Simulated Errors due to wavenumber scale uncertainties:

290 SHIS obs 12/13 Sept 1999, Roi Namur overpass 285 calculation Atmospheric CO₂ lines 280 275 from S-HIS and from 270 £_265 using HITRAN2000. 260 255 250 245 724-738 cm⁻¹ integrated residual (arb. units) 01 01 240 734 722 724 726 728 730 732 wavenumber (1/cm) **Observed** wavenumber scale factor chosen to minimize difference with radiative transfer Calculation. 10 0.9996 0.9997 0.9998 0.9999 1 1.0001 1.0002 1.0003 1.0004

laser wavenumber / (15799.6 cm⁻¹)

Off-Axis GIFTS Focal Plane Array Angle

64 κ. 民 60 45 . Ф 56 52 40 48 æ 44 40 30 ÷ 75 ⁼P.A. pixel index, j 36 32 iæ. 28 24 Τ. 20 권 16 12 g 8 4 炆 40 12 16 20 24 29 32 36 40 44 49 52 56 60 64 4 8 FPA pixel index, i

FPA off–axis angle,θ (milli⊢radians)

Self-Apodization due to finite detector size

Self-apodization effect is less than 1% for GIFTS detectors.

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Equivalent temperature error less than 0.1K even without correction!

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• In expression for the measured interferogram, F(x), expand sinc function as a power series of $(2\pi v x b\theta)$:

$$F(x) = \int d\mathbf{n} N(\mathbf{n}) e^{i2\mathbf{p}\mathbf{n}} - \frac{(2\mathbf{p}xb\mathbf{q})^2}{3!} \cdot \int d\mathbf{n} N(\mathbf{n}) \mathbf{n}^2 e^{i2\mathbf{p}\mathbf{n}} + \frac{(2\mathbf{p}xb\mathbf{q})^4}{5!} \cdot \int d\mathbf{n} N(\mathbf{n}) \mathbf{n}^4 e^{i2\mathbf{p}\mathbf{n}} - \dots$$

- Compute perturbation terms and subtract from measured interferogram.
- » Similar process is currently performed for UW FTS airborne and ground based instruments (S-HIS/AERI).

- The beams of light reaching each detector in the focal plane array pass through the interferometer at different angles, ϕ .
- With respect to the on-axis beam, the off-axis beams have slightly shorter OPDs:
 OPD(φ) = OPD(0) cos(φ).
- The varying OPD scale across the focal plane array leads to slightly different wavenumber scales for each detector in the focal plane array.
- Ground processing will be used to interpolate the radiance spectrum from each pixel onto a reference wavenumber scale.

from Fourier Transform Spectrometry, James W. Brault

GIFTS Off-Axis Interferogram Sampling

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Image of interferogram point #780 (CO₂ resonance)

Image of interferogram point #1 (ZPD)

- Primary issues to be addressed by the spectral calibration are ILS changes and wavenumber scale "stretch" across the array.
- ILS effects are negligible because of the small angular size of individual pixels.
- The wavenumber scale stretch variation over the array is large, but is physically well understood and can easily be removed.
- The spectral calibration will be routinely monitored using atmospheric emission lines during flight.
- Real-time ground data processing will produce a continuous stream of spectrally calibrated radiance observations on a standard wavenumber scale.

